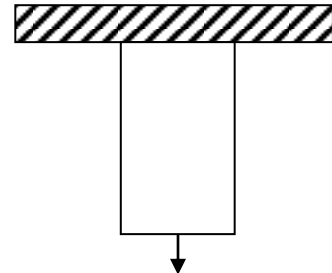


FEM
Module One Problems
By V Balaraj

1. A rectangular bar is subjected to an axial load P as shown in Figure. Determine the expression of PE functional and determine the extreme value of the PE for the following data E= 200 GPa, P = 3kN, L=100 mm, b=width = 20 mm and t = 10 mm .



Solution

PE= SE + WP

$$SE = \frac{1}{2} \sigma \epsilon V = \frac{1}{2} \sigma \epsilon A L$$

$$\epsilon = \frac{u}{L} \quad \text{and} \quad \sigma = E \epsilon = \frac{Eu}{L}$$

$$SE = \frac{1}{2} \frac{Eu^2 AL}{L^2} = \frac{1}{2} \frac{Eu^2 A}{L}$$

$$WP = - p u$$

$$\pi = \frac{1}{2} \frac{Eu^2 A}{L} - p u$$

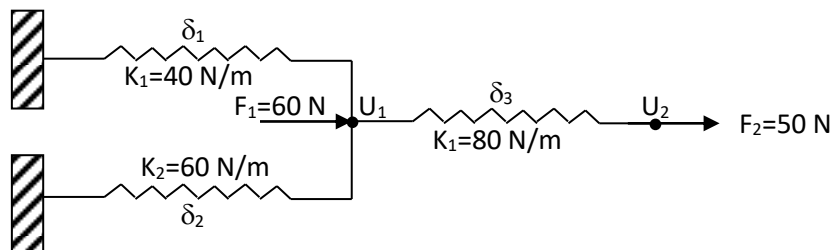
Apply minimum PE principle

$$\frac{\partial \pi}{\partial u} = \frac{Eu}{L} A - p = 0$$

$$u = \frac{3000 \times 100}{10 \times 20 \times 200 \times 10^3} = 7.5 \times 10^{-3} \text{ mm}$$

$$\pi = \frac{1}{2} \frac{Eu^2 A}{L} - p u = -11.25 \text{ N} - \text{ mm}$$

2. Using the principle of minimum potential energy determine the displacement at the nodes for a given spring system shown in figure.



$$PE = SE + WP$$

$$SE = \frac{1}{2}k_1\delta_1^2 + \frac{1}{2}k_2\delta_2^2 + \frac{1}{2}k_3\delta_3^2$$

$$SE = \frac{1}{2}k_1u_1^2 + \frac{1}{2}k_2u_2^2 + \frac{1}{2}k_3(u_2 - u_1)^2$$

$$WP = -F_1u_1 - F_2u_2$$

$$\pi = \frac{1}{2}k_1u_1^2 + \frac{1}{2}k_2u_2^2 + \frac{1}{2}k_3(u_2 - u_1)^2 - F_1u_1 - F_2u_2$$

$$\frac{\partial \pi}{\partial u_1} = k_1u_1 + k_2u_1 - k_3(u_2 - u_1) - F_1 = 0$$

$$(k_1 + k_2 + k_3)u_1 - k_3u_2 = F_1$$

$$180 u_1 - 80u_2 = 60 \quad \quad \quad \mathbf{01}$$

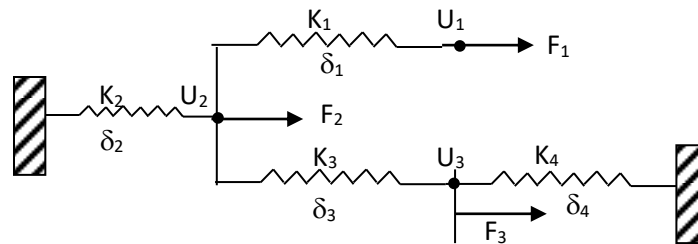
$$\frac{\partial \pi}{\partial u_2} = -k_3u_1 + k_3u_2 - F_2 = 0$$

$$-k_3u_1 + k_3u_2 = F_2$$

$$-80 u_1 + 80u_2 = 50 \quad \quad \quad \mathbf{02}$$

$$U_1 = 1.1 \text{ m and } U_2 = 1.725 \text{ m}$$

3. Using the principle of minimum potential energy determine the displacement at the nodes for a given spring system shown in figure.



$$PE = SE + WP$$

$$SE = \frac{1}{2}k_1\delta_1^2 + \frac{1}{2}k_2\delta_2^2 + \frac{1}{2}k_3\delta_3^2 + \frac{1}{2}k_4\delta_4^2$$

$$SE = \frac{1}{2}k_1(u_1 - u_2)^2 + \frac{1}{2}k_2u_2^2 + \frac{1}{2}k_3(u_3 - u_2)^2 + \frac{1}{2}k_4u_3^2$$

$$WP = -F_1u_1 - F_2u_2 - F_3u_3$$

$$\pi = \frac{1}{2}k_1(u_1 - u_2)^2 + \frac{1}{2}k_2u_2^2 + \frac{1}{2}k_3(u_3 - u_2)^2 + \frac{1}{2}k_4u_3^2 - F_1u_1 - F_2u_2 - F_3u_3$$

$$\frac{\partial \pi}{\partial u_1} = k_1(u_1 - u_2) - F_1 = 0$$

$$k_1u_1 - k_1u_2 = F_1 \quad 01$$

$$\frac{\partial \pi}{\partial u_2} = -k_1(u_1 - u_2) + k_2u_2 - k_3(u_3 - u_2) - F_2 = 0$$

$$-k_1u_1 + k_1u_2 + k_2u_2 + k_3u_2 - k_3u_3 = F_2$$

$$-k_1u_1 + (k_1 + k_2 + k_3)u_2 - k_3u_3 = F_2 \quad 02$$

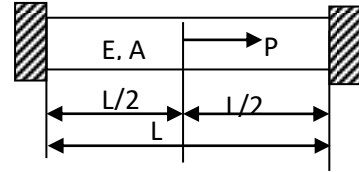
$$\frac{\partial \pi}{\partial u_3} = k_3(u_3 - u_2) + k_4u_3 - F_3 = 0$$

$$-k_3u_2 + (k_3 + k_4)u_3 = F_3 \quad 03$$

From Equation 1, 2 and 3

$$\begin{bmatrix} k_1 & -k_1 & 0 \\ -k_1 & k_1 + k_2 + k_3 & -k_3 \\ 0 & -k_3 & k_3 + k_4 \end{bmatrix} \begin{bmatrix} u_1 \\ u_2 \\ u_3 \end{bmatrix} = \begin{bmatrix} F_1 \\ F_2 \\ F_3 \end{bmatrix}$$

4. Figure shows a bar fixed at both ends subjected to an axial load. Determine the displacement at the loading point and the corresponding stress using Rayleigh- Ritz method.



Step 1: Formulation of PE functions

$$\begin{aligned} \Pi &= SE + WP \\ \Pi &= \frac{EA}{2} \int_0^L \left(\frac{\partial u}{\partial x} \right)^2 dx - p u_m \end{aligned} \quad 01$$

Step 2: Selection of Displacement Model

$$u = a_0 + a_1x + a_2x^2 \quad 02$$

$$\text{BC-1 } x = 0, u = 0 \text{ then } a_0 = 0$$

$$\text{BC-2 } x = L, u = 0 \rightarrow 0 = a_1L + a_2L^2$$

$$a_1 = -a_2L$$

$$u = -a_2Lx + a_2x^2 \rightarrow u = a_2(x^2 - Lx) \quad 03$$

$$\frac{\partial u}{\partial x} = a_2(2x - L)$$

$$\left(\frac{\partial u}{\partial x}\right)^2 = a_2^2(4x^2 + L^2 - 4Lx) \quad 04$$

$$x = \frac{L}{2} \rightarrow \rightarrow \rightarrow u = u_m = a_2 \left(\frac{L^2}{4} - \frac{L^2}{2}\right) \rightarrow \rightarrow u_m = -a_2 \frac{L^2}{4} \quad 05$$

Step 3: Substitute in PE

$$\Pi = \frac{EA}{2} \int_0^L a_2^2(4x^2 + L^2 - 4Lx) dx + p a_2 \frac{L^2}{4}$$

$$\Pi = \frac{E a_2^2 L^3}{6} + \frac{p a_2 L^2}{4} \quad 06$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial a_2} = \frac{2E a_2 L^3}{6} + \frac{p L^2}{4} = 0$$

$$a_2 = -\frac{3p}{4EAL} \quad 07$$

Substitute in Equation 03

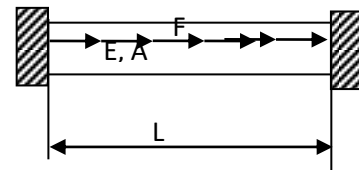
$$u = -\frac{3p}{4EAL}(x^2 - Lx) \quad 08$$

$$x = \frac{L}{2} \quad u = u_m = -\frac{3pL}{16EA}$$

Position (x)	Strain = $\frac{\partial u}{\partial x} = -\frac{3p}{4EAL}(2x - L)$	Stress = $-\frac{3p}{4AL}(2x - L)$
x = 0	$\frac{3p}{4EA}$	$\frac{3p}{4A}$
x = L/2	0	0
x = L	$-\frac{3p}{4EA}$	$-\frac{3p}{4A}$

5. Figure shows a bar fixed at both ends subjected to continuous axial load.

Determine the displacement at the loading point and the corresponding stress using Rayleigh- Ritz method.



Step 1: Formulation of PE functions

$$\Pi = SE + WP$$

$$\Pi = \frac{EA}{2} \int_0^L \left(\frac{\partial u}{\partial x}\right)^2 dx - \int_0^L F u dx \quad 01$$

Step 2: Selection of Displacement Model

$$u = a_0 + a_1x + a_2x^2 \quad 02$$

$$\text{BC-1 } x = 0, u=0 \text{ then } a_0=0$$

$$\text{BC-2 } x = L, u = 0 \rightarrow 0 = a_1L + a_2L^2$$

$$a_1 = -a_2L$$

$$u = -a_2Lx + a_2x^2 \rightarrow u = a_2(x^2 - Lx) \quad 03$$

$$\frac{\partial u}{\partial x} = a_2(2x - L)$$

$$\left(\frac{\partial u}{\partial x}\right)^2 = a_2^2(4x^2 + L^2 - 4Lx) \quad 04$$

$$\int_0^L Fa_2(x^2 - Lx)dx = -\frac{Fa_2L^3}{6} \quad 05$$

Step 3: Substitute in PE

$$\Pi = \frac{EA}{2} \int_0^L a_2^2(4x^2 + L^2 - 4Lx)dx + \frac{Fa_2L^3}{6}$$

$$\Pi = \frac{E a_2^2 L^3}{6} + \frac{F a_2 L^3}{6} \quad 06$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial x} = \frac{2E a_2 L^3}{6} + \frac{F L^3}{6} = 0$$

$$a_2 = -\frac{F}{2EA} \quad 07$$

Substitute in Equation 03

$$u = -\frac{F}{2EA}(x^2 - Lx) \quad 08$$

$$x = \frac{L}{2} \quad u = u_m = -\frac{3FL^2}{8EA}$$

Position (x)	Strain = $\frac{\partial u}{\partial x} = -\frac{F}{2EA}(2x - L)$	Stress = $-\frac{F}{2A}(2x - L)$
x = 0	$\frac{FL}{2EA}$	$\frac{FL}{2A}$
x=L/2	0	0
x = L	$-\frac{FL}{2EA}$	$-\frac{FL}{2A}$

6. A cantilever beam of span L is subjected to a point load at free end. Derive an equation for the deflection at free end by using Rayleigh- Ritz method. Assume polynomial function



Step 1: Formulation of PE functions

$$\Pi = SE + WP$$

$$\Pi = \frac{EI}{2} \int_0^L \left(\frac{\partial^2 y}{\partial x^2} \right)^2 dx - py_m \quad 01$$

Step 2: Selection of Displacement Model

$$y = a_0 + a_1x + a_2x^2 + a_3x^3 \quad 02$$

$$\text{BC-1 } x = 0, y = 0 \text{ then } a_0 = 0$$

$$\text{BC-2 } x = 0, \frac{\partial y}{\partial x} = 0 \text{ then } a_1 = 0$$

$$\text{BC-3 } x = L, \frac{\partial^2 y}{\partial x^2} = 0 \quad 0 = 2a_2 + 6a_3L$$

$$a_2 = -3a_3L$$

$$y = -3a_3Lx^2 + a_3x^3 \rightarrow y = a_3(x^3 - 3Lx^2) \quad 03$$

$$\frac{\partial y}{\partial x} = a_3(3x^2 - 6Lx)$$

$$\frac{\partial^2 y}{\partial x^2} = 6a_3(x - L)$$

$$\left(\frac{\partial^2 y}{\partial x^2} \right)^2 = 36a_3^2(x^2 + L^2 - 2Lx) \quad 04$$

$$x = L \rightarrow y = y_m = a_3(L^3 - 3L^3) \rightarrow Y_m = -2a_3L^3 \quad 05$$

Step 3: Substitute in PE

$$\Pi = \frac{EI}{2} \int_0^L 36a_3^2(x^2 + L^2 - 2Lx) + 2a_3L^3P$$

$$\Pi = 6EIa_3^2L^3 + 2a_3L^3P \quad 06$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial a_3} = 12EIa_3L^3 + 2L^3P = 0$$

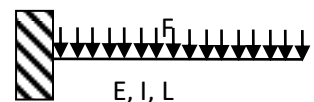
$$a_3 = -\frac{P}{6EI} \quad 07$$

Substitute in Equation 03

$$y = -\frac{P}{6EI}(x^3 - 3Lx^2) \quad 08$$

$$y_m = \frac{PL^3}{3EI}$$

7. A cantilever beam of span L is subjected to a uniformly distributed load. Derive an equation for the deflection at free end by using Rayleigh- Ritz method. Assume



polynomial function

Step 1: Formulation of PE functions

$$\Pi = SE + WP$$

$$\Pi = \frac{EI}{2} \int_0^L \left(\frac{\partial^2 y}{\partial x^2} \right)^2 dx - \int_0^L Fy dx \quad 01$$

Step 2: Selection of Displacement Model

$$y = a_0 + a_1x + a_2x^2 + a_3x^3 \quad 02$$

$$\text{BC-1 } x = 0, y=0 \text{ then } a_0=0$$

$$\text{BC-2 } x = 0, \frac{\partial Y}{\partial x} = 0 \text{ then } a_1=0$$

$$\text{BC-3 } x=L \frac{\partial^2 Y}{\partial x^2} = 0 \quad 0 = 2a_2 + 6a_3L$$

$$a_2 = -3a_3L$$

$$y = -3a_3Lx^2 + a_3x^3 \rightarrow y = a_3(x^3 - 3Lx^2) \quad 03$$

$$\frac{\partial y}{\partial x} = a_3(3x^2 - 6Lx)$$

$$\frac{\partial^2 y}{\partial x^2} = 6a_3(x - L)$$

$$\left(\frac{\partial^2 y}{\partial x^2} \right)^2 = 36a_3^2(x^2 + L^2 - 2Lx) \quad 04$$

$$\int_0^L Fa_3(x^3 - 3Lx^2) dx = -\frac{3Fa_3L^4}{4} \quad 05$$

Sep 3: Substitute in PE

$$\Pi = \frac{EI}{2} \int_0^L 36a_3^2(x^2 + L^2 - 2Lx) + \frac{3Fa_3L^4}{4}$$

$$\Pi = 6EIa_3^2L^3 + \frac{3Fa_3L^4}{4} \quad 06$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial x} = 12EIa_3L^3 + \frac{3FL^4}{4} = 0$$

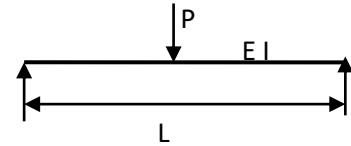
$$a_3 = -\frac{FL}{16EI} \quad 07$$

Substitute in Equation 03

$$y = -\frac{FL}{16EI}(x^3 - 3Lx^2) \quad 08$$

$$y_m = \frac{FL^4}{8EI}$$

8. Simply supported beam subjected to point load at the centre. Derive an equation for maximum deflection using trigonometrical function by Rayleigh Ritz method



Step 1: Formulation of PE functions

$$\Pi = SE + WP$$

$$\Pi = \frac{EI}{2} \int_0^L \left(\frac{\partial^2 y}{\partial x^2} \right)^2 dx - py_m \tag{01}$$

Step 2: Selection of Displacement Model

$$y = C \sin \frac{\pi x}{L} \tag{02}$$

BC-1 $x = 0$ then $y = 0$

BC-2 $x = L$ then $y = 0$

BC-3 $x = L/2$ then $\frac{\partial y}{\partial x} = 0$

$$y = C \sin \frac{\pi x}{L} \tag{03}$$

$$\frac{\partial^2 y}{\partial x^2} = -C \frac{\pi^2}{L^2} \sin \frac{\pi x}{L}$$

$$\left(\frac{\partial^2 y}{\partial x^2} \right)^2 = C^2 \frac{\pi^4}{L^4} \sin^2 \frac{\pi x}{L} \tag{04}$$

$$x = \frac{L}{2} \rightarrow \rightarrow \rightarrow y = y_m = C \tag{05}$$

Step 3: Substitute in PE

$$\Pi = \frac{EI}{2} \int_0^L C^2 \frac{\pi^4}{L^4} \sin^2 \frac{\pi x}{L} dx - Cp$$

$$\Pi = \frac{C^2 EI \pi^4}{4L^3} - pC \tag{06}$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial C} = \frac{2C EI \pi^4}{4L^3} - p = 0$$

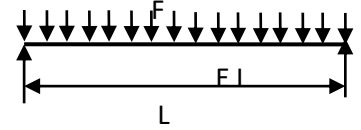
$$C = \frac{2PL^3}{EI \pi^4} \tag{07}$$

Substitute in Equation 03

$$y = \frac{2PL^3}{EI \pi^4} \sin \frac{\pi x}{L} \tag{08}$$

$$Y = \frac{2PL^3}{EI \pi^4}$$

9. Simply supported beam subjected uniform distributed load on the beam. Derive an equation for maximum deflection using trigonometrical function by Rayleigh Ritz method



Step 1: Formulation of PE functions

$$\Pi = SE + WP$$

$$\Pi = \frac{EI}{2} \int_0^L \left(\frac{\partial^2 y}{\partial x^2} \right)^2 dx - \int_0^L F y dx \tag{01}$$

Step 2: Selection of Displacement Model

$$y = C \sin \frac{\pi x}{L} \tag{02}$$

BC-1 $x = 0$ then $y = 0$

BC-2 $x = L$ then $y = 0$

BC-3 $x = L/2$ then $\frac{\partial y}{\partial x} = 0$

$$y = C \sin \frac{\pi x}{L} \tag{03}$$

$$\frac{\partial^2 y}{\partial x^2} = -C \frac{\pi^2}{L^2} \sin \frac{\pi x}{L}$$

$$\left(\frac{\partial^2 y}{\partial x^2} \right)^2 = C^2 \frac{\pi^4}{L^4} \sin^2 \frac{\pi x}{L} \tag{04}$$

$$x = \frac{L}{2} \rightarrow \rightarrow \rightarrow y = y_m = C \tag{05}$$

$$\int_0^L p y dx = \int_0^L p C \sin \frac{\pi x}{L} dx = \frac{2FCL}{\pi}$$

Step 3: Substitute in PE

$$\Pi = \frac{EI}{2} \int_0^L C^2 \frac{\pi^4}{L^4} \sin^2 \frac{\pi x}{L} - \frac{2FCL}{\pi}$$

$$\Pi = \frac{C^2 EI \pi^4}{4L^3} - \frac{2FCL}{\pi} \tag{06}$$

Step 4: Applied Minimum Potential Energy

$$\frac{\partial \Pi}{\partial C} = \frac{2C EI \pi^4}{4L^3} - \frac{2FL}{\pi} = 0$$

$$C = \frac{4FL^4}{EI \pi^5} \tag{07}$$

Substitute in Equation 03

$$y = \frac{4FL^4}{EI\pi^5} \sin \frac{\pi x}{L}$$

08

$$Y_m = \frac{4FL^4}{EI\pi^5} = \frac{FL^4}{76.52 EI}$$

10. A cantilever beam of span L is subjected to a uniformly distributed load. Derive an equation for the deflection at free end by using Galerkin technique. Assume polynomial function.



Step 1: Formulation of DE functions

$$EI \frac{\partial^4 y}{\partial x^4} - F = 0 \quad 01$$

Step 2: Selection of Displacement Model

$$y = a_0 + a_1 x + a_2 x^2 + a_3 x^3 + a_4 x^4 \quad 02$$

BC-1 $x = 0, y = 0$ then $a_0 = 0$

BC-2 $x = 0, \frac{\partial y}{\partial x} = 0$ then $a_1 = 0$

BC-3 $x = L, \frac{\partial^2 y}{\partial x^2} = 0$ $0 = 2a_2 + 6a_3 L + 12a_4 L^2$

$$a_2 = -3a_3 L - 6a_4 L^2 \quad 03$$

BC-4 $x = L, \frac{\partial^3 y}{\partial x^3} = 0$ $0 = 6a_3 + 24 a_4 L$

$$a_3 = -4 a_4 L \quad 04$$

Substitute (4) into (3)

$$a_2 = -3(-4 a_4 L) L - 6a_4 L^2$$

$$a_2 = 12 a_4 L^2 - 6a_4 L^2$$

$$a_2 = 6 a_4 L^2 \quad 05$$

Substitute equation 4 and 5 in equation 1

$$y = 6 a_4 L^2 x^2 - 4 a_4 L x^3 + a_4 x^4$$

$$y = a_4 (x^4 - 4 L x^3 + 6 L^2 x^2) \quad 06$$

$$w_1 = x^4 - 4 L x^3 + 6 L^2 x^2$$

$$\frac{\partial y}{\partial x} = a_4 (4x^3 - 12 L x^2 + 12 L^2 x)$$

$$\frac{\partial^2 y}{\partial x^2} = a_4 (12x^2 - 24 L x + 12 L^2)$$

$$\frac{\partial^3 y}{\partial x^3} = a_4 (24x - 24 L)$$

$$\frac{\partial^4 y}{\partial x^4} = 24 a_4 \quad 07$$

Sep 3: Substitute in DE and equating to R

$$EI 24 a_4 - F = R \quad 08$$

Step 4: Applied Galerkin technique (weight residual technique)

$$\int_0^L w_1 R \, dx = 0 \quad 09$$

$$\int_0^L (x^4 - 4Lx^3 + 6L^2x^2)(EI \, 24a_4 - F) \, dx = 0$$

$$EI \, 24a_4 - F = 0$$

$$a_4 = \frac{F}{24EI} \quad 10$$

Substitute in equation 6

$$y = \frac{F}{24EI} (x^4 - 4Lx^3 + 6L^2x^2) \quad 11$$

Maximum deflection is at $x = L$

$$y = \frac{F}{24EI} (L^4 - 4LL^3 + 6L^2L^2)$$

$$y_m = \frac{F3L^4}{24EI}$$

$$y_m = \frac{FL^4}{8EI}$$

11. A cantilever beam of span L is subjected to a point load at free end. Derive an equation for the deflection at free end by using Galerkin method.



Assume polynomial function

Step 1: Formulation of DE functions

$$EI \frac{\partial^4 y}{\partial x^4} = 0 \quad 01$$

Step 2: Selection of Displacement Model

$$y = a_0 + a_1x + a_2x^2 + a_3x^3 + a_4x^4 \quad 02$$

$$\text{BC-1 } x = 0, y = 0 \quad \text{then } a_0 = 0$$

$$\text{BC-2 } x = 0, \frac{\partial y}{\partial x} = 0 \quad \text{then } a_1 = 0$$

$$\text{BC-3 } x = L, \frac{\partial^2 y}{\partial x^2} = 0 \quad 0 = 2a_2 + 6a_3L + 12a_4L^2$$

$$a_2 = -3a_3L - 6a_4L^2 \quad 03$$

$$\text{BC-4 } x = L, \frac{\partial^3 y}{\partial x^3} = \frac{-P}{EI} \quad \frac{-P}{EI} = 6a_3 + 24a_4L$$

$$a_3 = \frac{-P}{6EI} - 4a_4L \quad 04$$

Substitute (4) into (3)

$$a_2 = -3\left(\frac{-P}{6EI} - 4a_4L\right)L - 6a_4L^2$$

$$a_2 = \frac{PL}{2EI} + 12 a_4 L^2 - 6 a_4 L^2$$

$$a_2 = \frac{PL}{2EI} + 6 a_4 L^2 \quad 05$$

Substitute equation 4 and 5 in equation 1

$$y = \left(\frac{PL}{2EI} + 6 a_4 L^2\right)x^2 + \left(\frac{-P}{6EI} - 4 a_4 L\right)x^3 + a_4 x^4$$

$$y = \frac{PL}{2EI} x^2 + 6 a_4 L^2 x^2 - \frac{P}{6EI} x^3 - 4 a_4 L x^3 + a_4 x^4$$

$$y = \frac{P}{2EI} \left(Lx^2 - \frac{x^3}{3}\right) + a_4 (6L^2 x^2 - 4 Lx^3 + x^4) \quad 06$$

$$w_1 = 6L^2 x^2 - 4 Lx^3 + x^4$$

$$\frac{\partial y}{\partial x} = \frac{P}{2EI} \left(2Lx - \frac{3x^2}{3}\right) + a_4 (12L^2 x - 12 Lx^2 + 4x^3)$$

$$\frac{\partial^2 y}{\partial x^2} = \frac{P}{2EI} \left(2L - \frac{6x}{3}\right) + a_4 (12L^2 - 24 Lx + 12x^2)$$

$$\frac{\partial^3 y}{\partial x^3} = \frac{P}{2EI} \left(0 - \frac{6}{3}\right) + a_4 (24 L - 24x)$$

$$\frac{\partial^4 y}{\partial x^4} = 24a_4 \quad 07$$

Sep 3: Substitute in DE and equating to R

$$EI 24a_4 = R \quad 08$$

Step 4: Applied Galerkin technique (weight residual technique)

$$\int_0^1 w_1 R dx = 0 \quad 09$$

$$\int_0^1 (x^4 - 4 Lx^3 + 6L^2 x^2) 24a_4 dx = 0$$

$$a_4 = 0 \quad 10$$

Substitute in equation 6

$$y = \frac{P}{2EI} \left(Lx^2 - \frac{x^3}{3}\right) + 0(6L^2 x^2 - 4 Lx^3 + x^4)$$

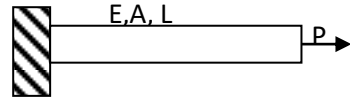
$$y = \frac{P}{2EI} \left(Lx^2 - \frac{x^3}{3} \right)$$

Maximum deflection is at $x = L$

$$y_m = \frac{P}{2EI} \left(LL^2 - \frac{L^3}{3} \right)$$

$$y_m = \frac{PL^3}{3EI}$$

12. Use Galerkin method to find the displacement of the system shows in Fig.



Step 1: Formulation of DE functions

$$EA \frac{\partial^2 u}{\partial x^2} = 0 \quad 01$$

Step 2: Selection of Displacement Model

$$u = a_0 + a_1 x + a_2 x^2 \quad 02$$

BC-1 $x = 0, y=0$ then $a_0=0$

BC-2 $x = L, \frac{\partial Y}{\partial x} = \frac{P}{AE} = a_1 + 2a_2 L$

$$a_1 = \frac{P}{AE} - 2a_2 L$$

Substitute in 2

$$u = \frac{Px}{AE} - a_2(x^2 - 2Lx)$$

$$u = \frac{Px}{AE} - a_2(x^2 - 2Lx) \quad 03$$

$$w_1 = x^2 - 2Lx$$

$$\frac{\partial u}{\partial x} = \frac{P}{AE} - a_2(2x - 2L)$$

$$\frac{\partial^2 u}{\partial x^2} = 2a_2$$

Sep 3: Substitute in DE and equating to R

$$2EAa_2 = R \quad 04$$

Step 4: Applied Galerkin technique (weight residual technique)

$$\int_0^L w_1 R \, dx = 0 \quad 05$$

$$\int_0^L (x^2 - 2Lx) \, 2EAa_2 \, dx = 0$$

$$a_2 = 0 \quad 06$$

Substitute in equation 6

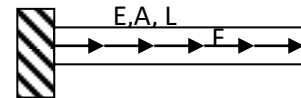
$$u = \frac{Px}{AE} - 0(x^2 - 2Lx)$$

$$u = \frac{Px}{AE}$$

Maximum deflection is at $x = L$

$$u_m = \frac{PL}{AE}$$

13. Use Galerkin method to find the displacement of the system shows in Fig.

**Step 1: Formulation of DE functions**

$$EA \frac{\partial^2 u}{\partial x^2} + F = 0 \quad 01$$

Step 2: Selection of Displacement Model

$$u = a_0 + a_1 x + a_2 x^2 \quad 02$$

$$\text{BC-1 } x = 0, y = 0 \quad \text{then } a_0 = 0$$

$$\text{BC-2 } x = L, \frac{\partial y}{\partial x} = 0 = a_1 + 2a_2 L$$

$$a_1 = -2a_2 L$$

Substitute in 2

$$u = a_2(x^2 - 2Lx) \quad 03$$

$$w_1 = x^2 - 2Lx$$

$$\frac{\partial u}{\partial x} = a_2(2x - 2L)$$

$$\frac{\partial^2 u}{\partial x^2} = 2a_2$$

Sep 3: Substitute in DE and equating to R

$$2EAa_2 + F = R \quad 04$$

Step 4: Applied Galerkin technique (weight residual technique)

$$\int_0^L w_1 R \, dx = 0 \quad 05$$

$$\int_0^L (x^2 - 2Lx) (2EAa_2 + F) \, dx = 0$$

$$(2EAa_2 + F) = 0 \quad 06$$

$$a_2 = -\frac{F}{2EA}$$

Substitute in equation 6

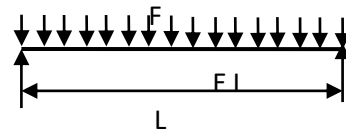
$$u = -\frac{F}{2EA} (x^2 - 2Lx)$$

$$u = \frac{F}{2EA} (2Lx - x^2)$$

Maximum deflection is at $x = L$

$$u_m = \frac{PL^2}{2AE}$$

14. A simply supported beam is subjected UDL as shown in Figure. Determine the maximum deflection using Galerkin method.



Step 1: Formulation of DE functions

$$EI \frac{\partial^4 y}{\partial x^4} - F = 0 \quad 01$$

Step 2: Selection of Displacement Model

$$y = C \sin \frac{\pi x}{L} \quad 02$$

$$w_1 = \sin \frac{\pi x}{L}$$

$$\frac{\partial y}{\partial x} = C \frac{\pi}{L} \cos \frac{\pi x}{L}$$

$$\frac{\partial^2 y}{\partial x^2} = -C \frac{\pi^2}{L^2} \sin \frac{\pi x}{L}$$

$$\frac{\partial^3 y}{\partial x^3} = -C \frac{\pi^3}{L^3} \cos \frac{\pi x}{L}$$

$$\frac{\partial^4 y}{\partial x^4} = C \frac{\pi^4}{L^4} \sin \frac{\pi x}{L}$$

Sep 3: Substitute in DE and equating to R

$$EI C \frac{\pi^4}{L^4} \sin \frac{\pi x}{L} - F = R$$

08

Step 4: Applied Galerkin technique (weight residual technique)

$$\int_0^L \sin \frac{\pi x}{L} (EI C \frac{\pi^4}{L^4} \sin \frac{\pi x}{L} - F) dx = 0$$

$$\int_0^L (EI C \frac{\pi^4}{L^4} \sin^2 \frac{\pi x}{L} - F \sin \frac{\pi x}{L}) dx = 0$$

$$\int_0^L (EI C \frac{\pi^4}{L^4} (\frac{1 - \cos \frac{2\pi x}{L}}{2}) - F \sin \frac{\pi x}{L}) dx = 0$$

$$EI C \frac{\pi^4}{L^4} \frac{L}{2} + F \frac{L}{\pi} (-1 - 1) = 0$$

$$EI C \frac{\pi^4}{2L^3} - \frac{2FL}{\pi} = 0$$

$$C = \frac{4FL^4}{EI\pi^5}$$

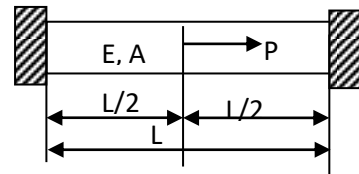
$$y = \frac{4FL^4}{EI\pi^5} \sin \frac{\pi x}{L}$$

Maximum deflection is at $x = L/2$

$$y = \frac{4FL^4}{EI\pi^5} \sin \frac{\pi}{2}$$

$$y_m = \frac{4FL^4}{EI\pi^5}$$

15. Figure shows a bar fixed at both ends subjected to an axial load. Determine the displacement at the loading point and the corresponding stress using Galerkin method. ($E=1$, $A=1$, $L=2$ and $p=2$)



$$\frac{d}{dx} EA \frac{\partial u}{\partial x} = 0 \quad \begin{array}{l} x=0 \text{ and } u=0 \\ x=L \text{ then } u=0 \end{array}$$

Multiply by ϕ (virtual displacement) and integrate by parts

$$\int_0^L \frac{d}{dx} EA \frac{\partial u}{\partial x} dx \phi = 0$$

$$\left[EA \frac{\partial u}{\partial x} \phi \right]_0^L - \int_0^L EA \frac{\partial u}{\partial x} \frac{\partial \phi}{\partial x} dx = 0$$

$$P\phi_1 - \int_0^L EA \frac{\partial u}{\partial x} \frac{\partial \phi}{\partial x} dx = 0$$

Assume trial function

$$u = a_0 + a_1 x + a_2 x^2 \quad 02$$

$$\text{BC-1 } x = 0, y = 0 \quad \text{then } a_0 = 0$$

$$\text{BC-2 } x = L, y = 0 = a_1 + 2a_2 L = 0$$

$$a_1 = -2a_2 L$$

$$u = -2a_2 Lx + a_2 x^2$$

$$u = a_2 (x^2 - Lx) = u_1 (x^2 - Lx)$$

$$\frac{\partial u}{\partial x} = u_1 (2x - L) \quad \text{Similarly for } \phi \quad 03$$

$$\frac{\partial \phi}{\partial x} = \phi_1 (2x - L) \quad 04$$

$$P\phi_1 - \int_0^L EA u_1 (2x - L) \phi_1 (2x - L) dx = 0$$

$$P\phi_1 - EA u_1 \int_0^L (4x^2 + L^2 - 4Lx) \phi_1 dx = 0$$

$$P\phi_1 - EA u_1 \left[\frac{4x^3}{3} + xL^2 - \frac{4Lx^2}{2} \right]_0^L \phi_1 = 0$$

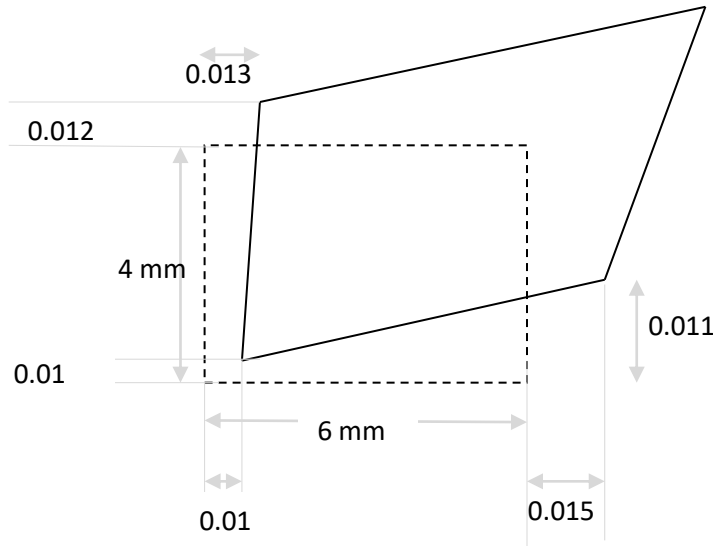
$$P\phi_1 - EA u_1 \left[\frac{4L^3}{3} + L^3 - \frac{L^3}{3} \right] \phi_1 = 0$$

$$P\phi_1 - EA u_1 \left[\frac{L^3}{3} \right] \phi_1 = 0$$

$$\phi_1 (P - EAu_1 \left[\frac{L^3}{3} \right]) = 0$$

$$u_1 = \frac{3P}{EAL^3} = \frac{3 \cdot 2}{1 \cdot 1 \cdot 8} = \frac{3}{4}$$

16. Estimate the three strains for given problem



$$\varepsilon_x = \frac{du}{dx} = \frac{0.015 - 0.01}{6} = 0.000833$$

$$\varepsilon_y = \frac{dv}{dy} = \frac{0.012 - 0.01}{4} = 0.0005$$

$$\gamma_{xy} = \frac{dv}{dx} + \frac{du}{dy} = \frac{0.011 - 0.01}{6} + \frac{0.013 - 0.01}{4} = 0.000917$$

17. Estimate the three strains and three stress for given problem. Assume given problem is plane stress method. ($E=210$ GPa, $\mu =0.3$) (continuation with previous problem)

$$\begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \end{Bmatrix} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix} \begin{Bmatrix} \varepsilon_x \\ \varepsilon_y \\ \gamma_{xy} \end{Bmatrix}$$

$$\begin{Bmatrix} \sigma_x \\ \sigma_y \\ \tau_{xy} \end{Bmatrix} = \frac{210e3}{1-0.3^2} \begin{bmatrix} 1 & 0.3 & 0 \\ 0.3 & 1 & 0 \\ 0 & 0 & \frac{1-0.3}{2} \end{bmatrix} \begin{Bmatrix} 0.0008333 \\ 0.0005 \\ 0.000917 \end{Bmatrix} = \begin{Bmatrix} 226.9154 \\ 173.0746 \\ 74.06538 \end{Bmatrix} MPa$$

18. Solve x, y and z values using Gauss elimination method for the following equation

$$\begin{aligned} 2x + 4y - 2z &= 2 \\ 4x + 9y - 3z &= 8 \\ -2x - 3y + 7z &= 10 \end{aligned}$$

$$\begin{aligned} 2x + 4y - 2z &= 2 \\ y + z &= 4 & R_2' \rightarrow R_2 - 2R_1 \\ y + 5z &= 12 & R_3' \rightarrow R_3 + R_1 \end{aligned}$$

$$\begin{aligned} 2x + 4y - 2z &= 2 \\ y + z &= 4 \\ 4z &= 8 & R_3'' \rightarrow R_3' - R_2' \end{aligned}$$

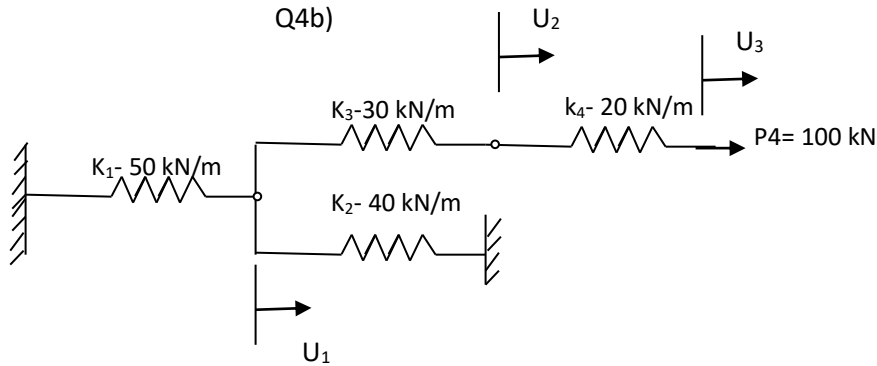
$$\begin{aligned} 2x + 4y - 2z &= 2 \\ y + z &= 4 \\ z &= 2 & \text{Divide by 2} \end{aligned}$$

$$\begin{aligned} 2x + 4y - 4 &= 2 \\ y + 2 &= 4 & \text{Substitute Z values} \\ z &= 2 \end{aligned}$$

$$\begin{aligned} 2x + 4y - 4 &= 2 \\ y &= 2 & \text{Y value obtained} \\ z &= 2 \end{aligned}$$

$$\begin{aligned} 2x + 8 - 4 &= 2 & \text{Substitute y and z values} \\ y &= 2 \\ z &= 2 \end{aligned}$$

$x = -1, y = 2$ and $z = 2$



$$PE = SE + WP$$

$$SE = \frac{1}{2}k_1\delta_1^2 + \frac{1}{2}k_2\delta_2^2 + \frac{1}{2}k_3\delta_3^2 + \frac{1}{2}k_4\delta_4^2$$

$$SE = \frac{1}{2}k_1u_1^2 + \frac{1}{2}k_2u_1^2 + \frac{1}{2}k_3(u_2 - u_1)^2 + \frac{1}{2}k_4(u_3 - u_2)^2$$

$$WP = -P_4u_3$$

$$\pi = \frac{1}{2}k_1u_1^2 + \frac{1}{2}k_2u_1^2 + \frac{1}{2}k_3(u_2 - u_1)^2 + \frac{1}{2}k_4(u_3 - u_2)^2 - P_4u_3$$

$$\frac{\partial \pi}{\partial u_1} = k_1u_1 + k_2u_1 + k_3u_1 - k_3u_2 = 0$$

$$\frac{\partial \pi}{\partial u_1} = (k_1 + k_2 + k_3)u_1 - k_3u_2 = 0$$

$$(50 + 40 + 30)u_1 - 30u_2 = 0$$

$$120u_1 - 30u_2 = 0 \text{ -----(1)}$$

$$\frac{\partial \pi}{\partial u_2} = k_3u_2 - k_3u_1 + k_4u_2 - k_4u_3 = 0$$

$$-k_3u_1 + (k_3 + k_4)u_2 - k_4u_3 = 0$$

$$-30u_1 + (30 + 20)u_2 - 20u_3 = 0$$

$$-30u_1 + 50u_2 - 20u_3 = 0 \text{ -----(2)}$$

$$\frac{\partial \pi}{\partial u_3} = -k_4 u_2 + k_4 u_3 - P_4 = 0$$

$$\frac{\partial \pi}{\partial u_3} = -20u_2 + 20u_3 = 100000 \text{-----}(3)$$

From 1, 2 and 3 we get

u3= 9444.444
u2= 4444.444
u1= 1111.111